



Propulsion Laboratory

National Aeronautics & Space Administration
California Institute of Technology



Reliability of Commercially Available Ball Grid Array Packages for Aerospace Applications



by

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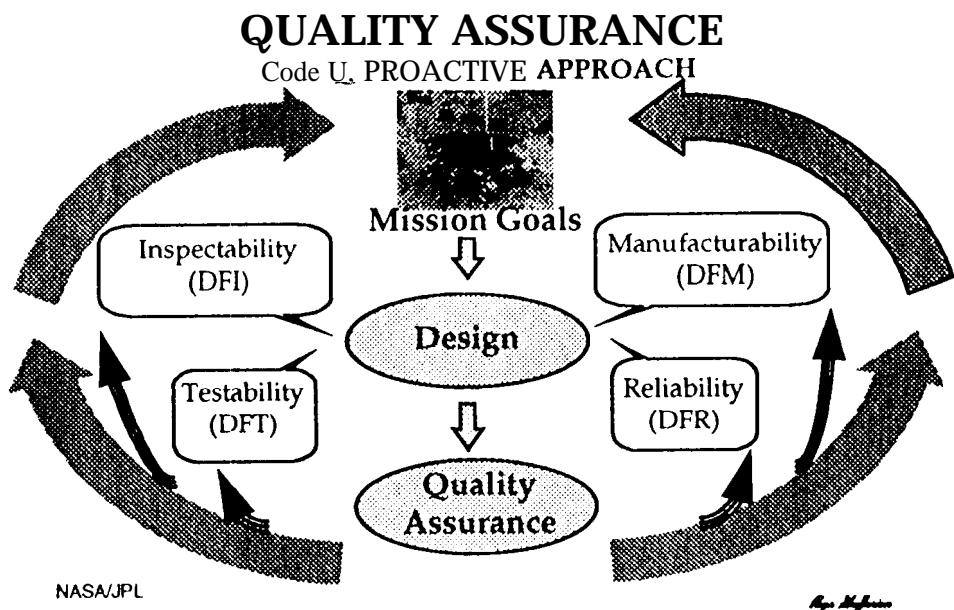
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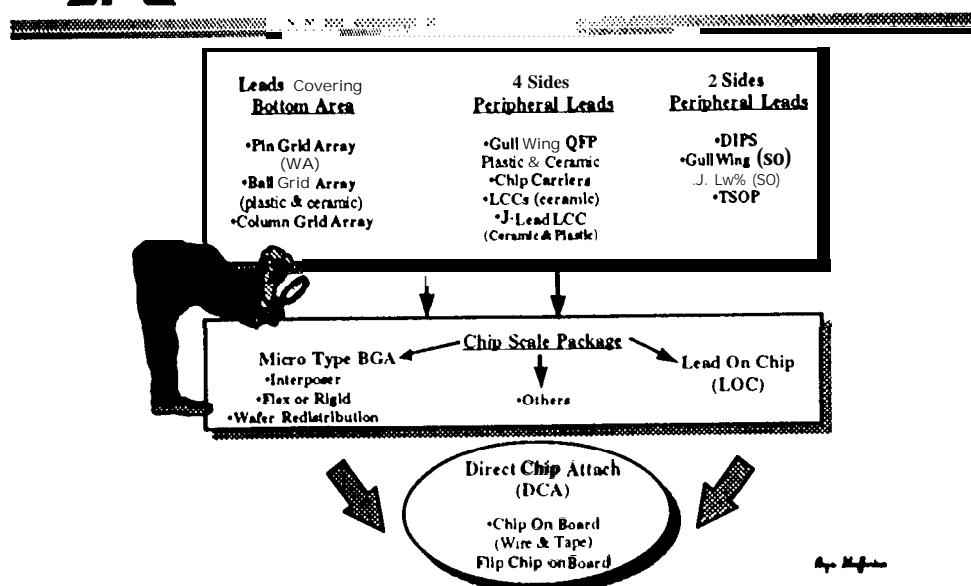
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OUTLINE

- QA Precative Approach
- SM Package Trends
- Introduction on Ball Grid Array
 - Why BGA
 - JPL-led BGA consortium
 - Test Matrix- Design of Experiment
 - Test Vehicle using commercially available BGA packages
- Environmental Test
 - Results on CBGA and PBGA
 - Failure Mechanisms
- Conclusions



SM PACKAGE MINIATURIZATION TRENDS



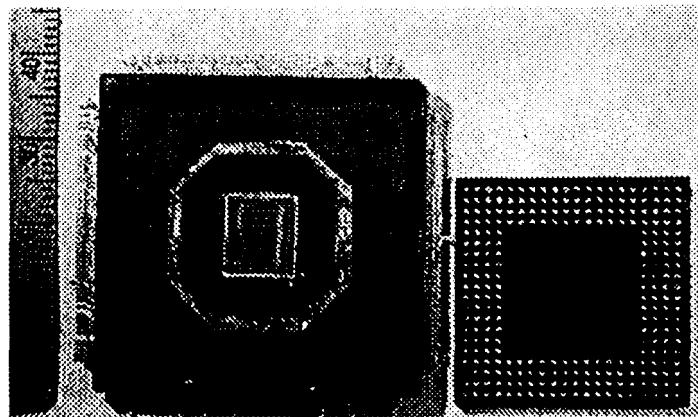
•Advantages

- o Capable of high pin counts
- o Manufacturing robustness
- o Higher package densities
- o Faster circuitry speed than QFP
- o Better heat dissipation

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•Challenges**o Inspection**

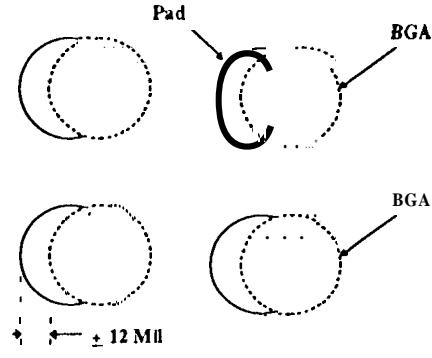
- o Multiple processes and double sided assemblies
- o Routing for high pin count
- o Rework



BGAS ARE ROBUST IN MANUFACTURING

BGAs Are Robust

- Placement accuracy: $\pm 12\text{ mil}$
- Vision Not Required
- BGA Self-Center during Reflow Process



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OBJECTIVES

Demonstrate robustness and quality & reliability of

Ball Grid Arrays (BGAs) interconnects

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Support the development of industrial infrastructure in
BGA product assurance

- Inspection methodology development, especially for assembly level
- Optimal package configuration- Full vs. peripheral
- Reliability characterization of CBGA/PBGA
 - I/O & Environmental dependency
- Reworking techniques

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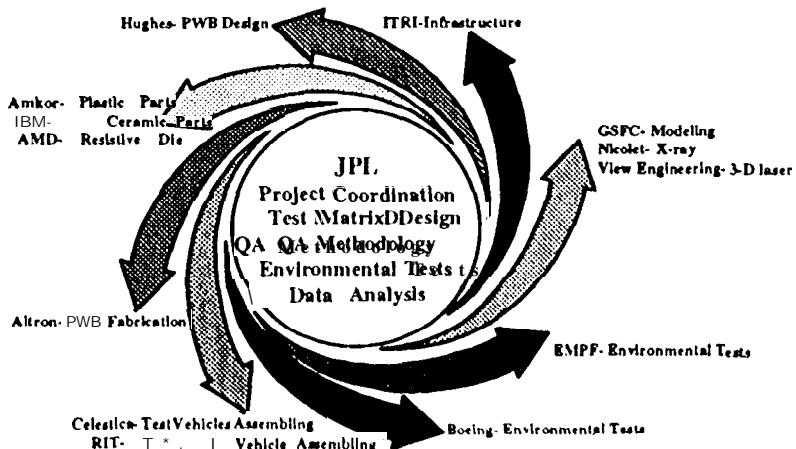
APPROACH

- Formed core consortium teams (JPL, Hughes, Boeing, RIT)
 - Reviewed literature and defined critical variables
- Workshop at JPL to understand industry's needs
 - Trends data from ITRI & SEMATECH, and input from others
 - Design of Experiment (DOE) test matrix used for efficiency
- Assemble more than 200 Test Vehicles
 - 300& 6001/0s packages, FR-4, Polyimide, 3 solder volume, 2 flux types, 2 manufacturing sites, 2 temp. cycling ranges, power cycling, and limited other variables

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CONSORTIUM



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JPL DESIGN OF EXPERIMENT

• 2- Level toggled parameters

- o Package type - CBGA and PBGA
- o Package 1/0s - "300" and "600"
- o PWB materials - FR-4 and Polyimide
- Flux type - RMA and Water soluble
- o Environmental test- two temperature ranges
- o Manufacturing sites- Mature (Celestica) and startup (RiT)

• 3- Level toggled parameters

- O Solder volume
- o Surface finish- OSP, limited HASL and electroless Ni/Au immersion

• Others

- o Power cycling and vibration toggled with PWB materials
- o Mixed technologies- Manufacturing robustness of BGA and 16 mil pitch QFP

JPL PACKAGE/ASSEMBLY CRITICAL VARIABLES

• BGA PACKAGE TYPE& I/O

- o Package type - CBGA and PBGA with die
- o Package 1/0s - "300" and "600", Package planarity
- o PWB materials - FR-4 and Polyimide

• MANUFACTURING

- o Process optimization
- o Mixed technology- Robustness of BGA and 16 mil QFP
- o Solder volume, Flux type - RMA, Water soluble, no clean
- o Surface finish- OSP, HASL and electroless Ni/Au immersion

• ENVIRONMENTAL CONDITION

- o Temperature cycle
- o Power cycle, vibration, etc.

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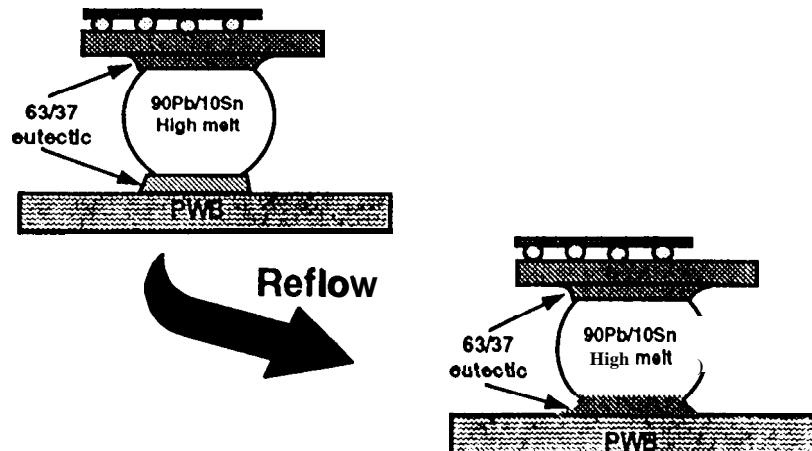
PACKAGE PLANARITY

- CBGA625 <.004" & CBGA361< .002
 - Coplanarity is critical
 - High melt solder ball and no collapse in reflow
- SuperBGA 560 & 352<.006 or higher
- OMPAC<.006
 - Coplanarity is not as critical as CBGA
 - Solder balls collapse during reflow
- 3-D Laser
 - Fast and accurate for planarity, warpage, and through position measurements

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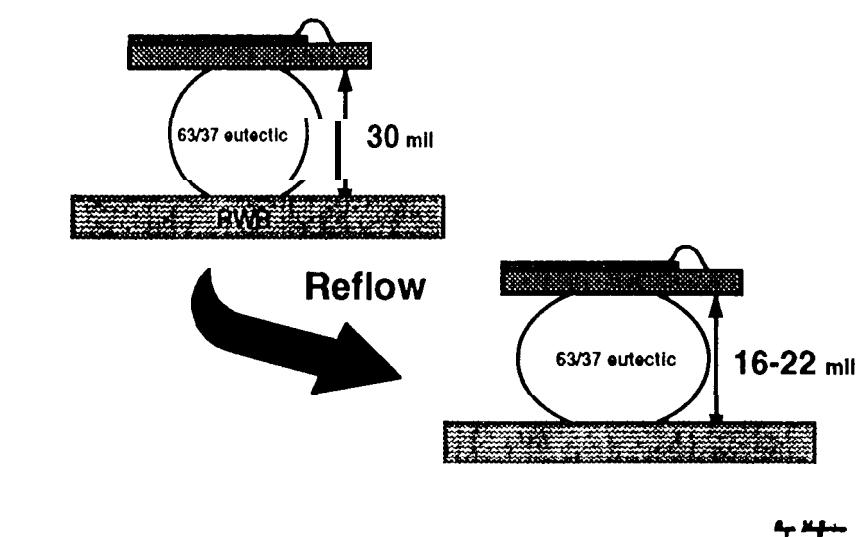
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CERAMIC BGA SOLDER BALLS CONTROL HEIGHT

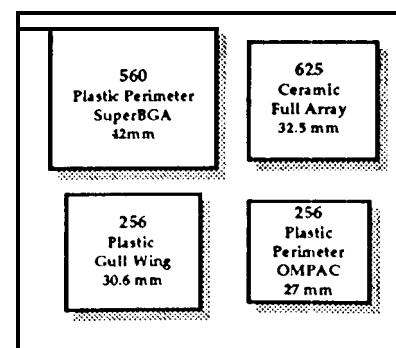
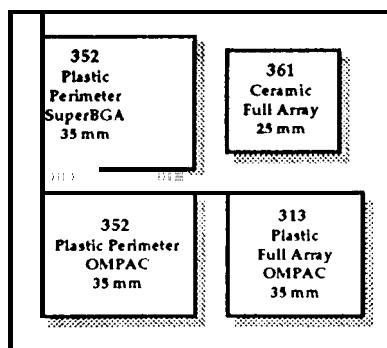


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PLASTIC BGA SOLDER BALLS COLLAPSE



JPL TEST VEHICLES



- THERMAL CYCLING
 - Military (-55 to 125 °C)
 - Others, majority (-30 to 100 °C), few NASA cycle (-55 to 100 °C) for baseline
- . TESTING TIME (estimated using Motorola data)
 - Type 1 -CBGA 361 for Boeing cycle- low -400 and No- 800 cycles
PBGA 361- low -1800 and average 3,255
 - Type 2- CBGA 625- low 150, No- 200

Due to larger CTE mismatch of CBGA packages on FR-4/Polyimide, time to failures for CBGA assemblies are much shorter than PBGAs

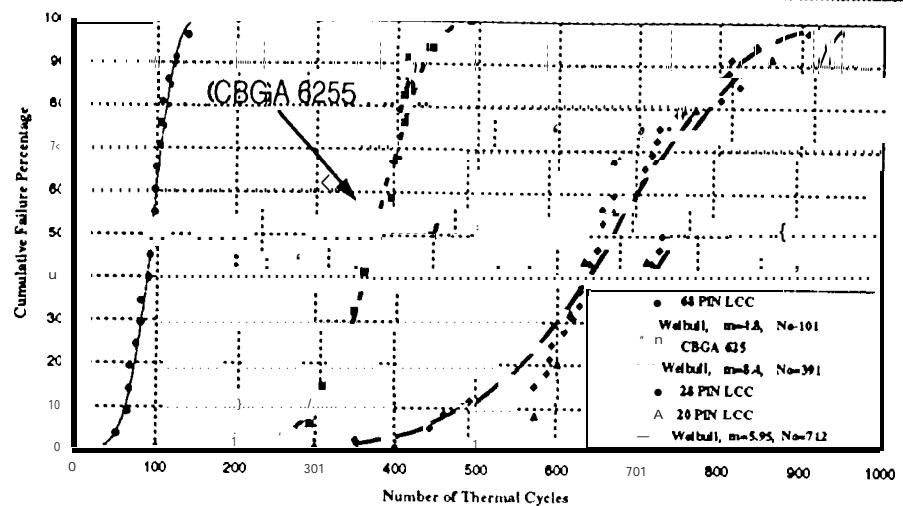


- Weibull Reliability Plots
 - Results for DOE matrix variables
- Coffin-Manson Relationship
 - Increase in package I/O & Change in temperature ranges
- Manufacturing defect
 - Frequency of defects for BGA and mixed technology
- Correlation
 - Mfg. defect and cycles to failure and crack propagation mapping
- Guidelines Documentation



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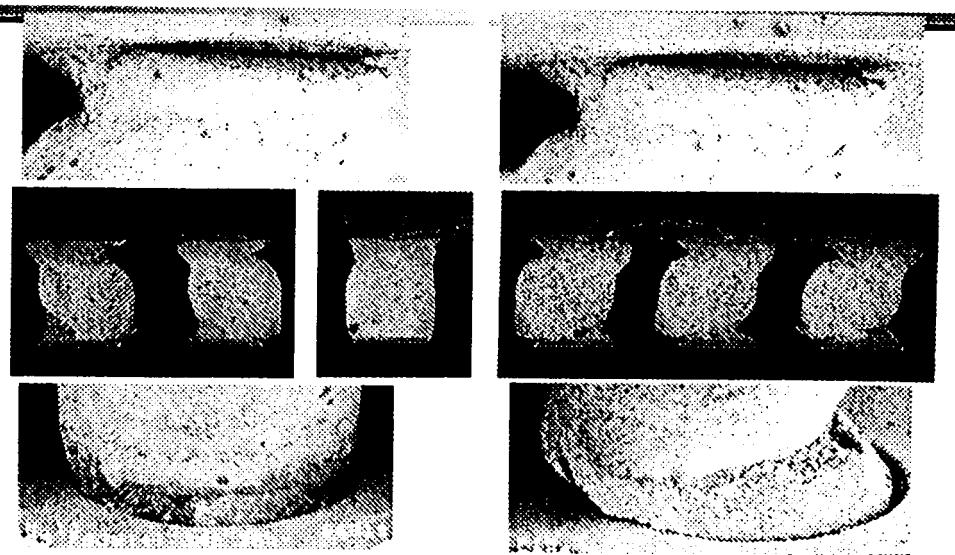
FAILURE DISTRIBUTIONS CBGA 625 & 20-, 28-, 68-PIN LCCS



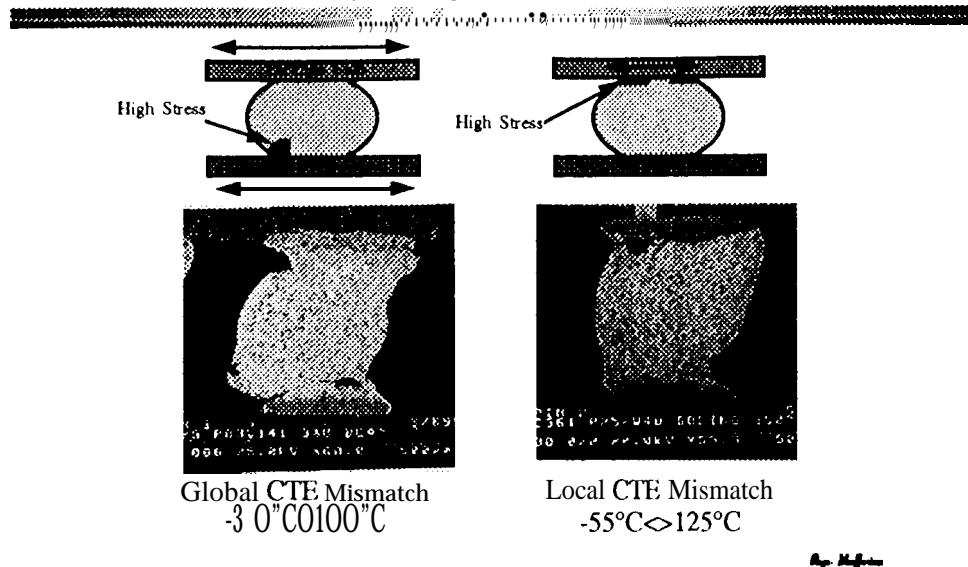
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Failure in CBGA 625

(-30°C to 100°C)

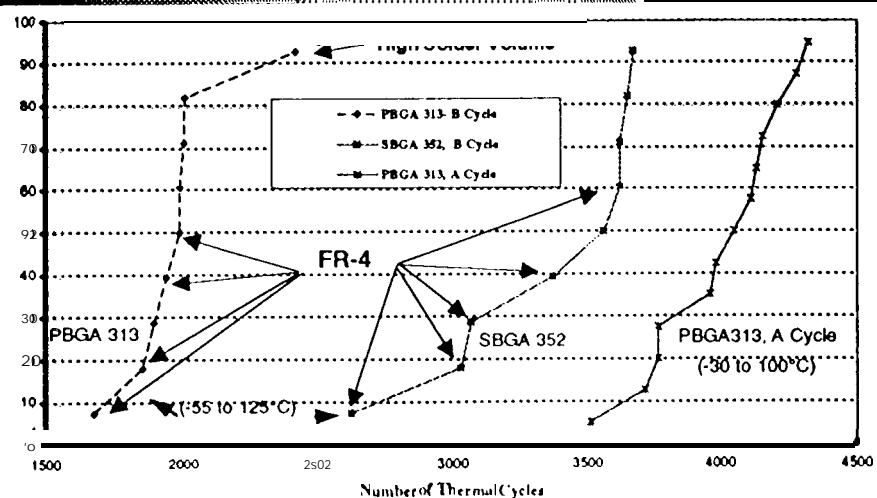


Failure Mechanism JPL Cycling Condition



FAILURE DISTRIBUTION

JPL PBGAs- (55°C to 125°C , 6S minutes, B; -30°C to 100°C , 82 minutes, A)



CONCLUSIONS

- CBGA 6251/0s were first to fail under different cycling ranges than
 - CBGA361
 - SBGA 560, SBGA 352, OMPAC 352, and PBGA 256 when cycled to different temperature ranges
 -
- PBGAs 313 **1/0**, depopulated full arrays, were first among the PBGAs to fail
 -
- SBGA 352 with no solder balls under the die showed much higher cycles to failure than the PBGA 313

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4-4-